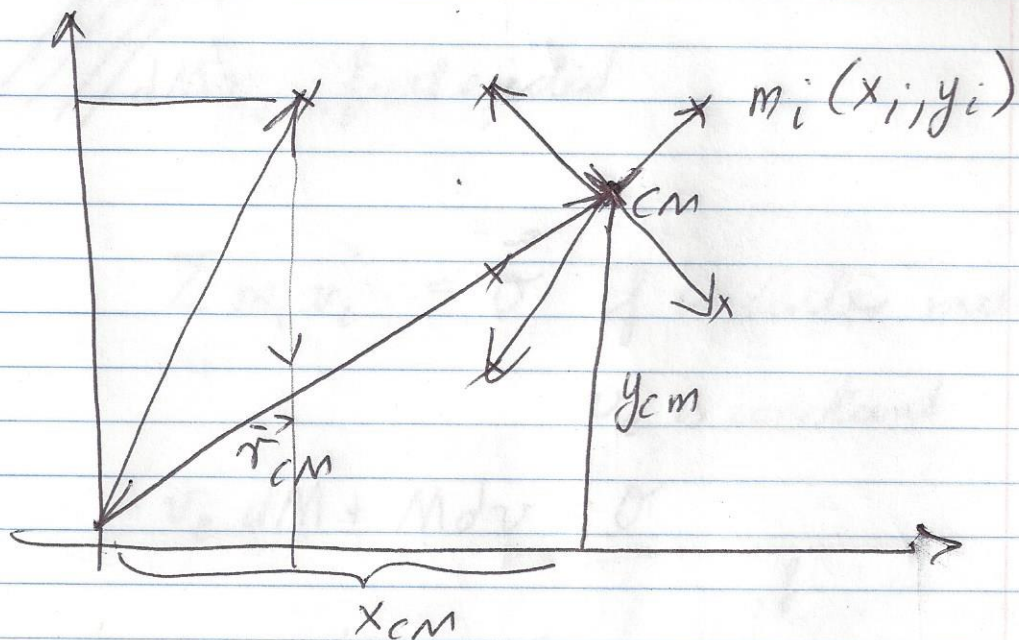


Point masses  $\rightarrow$  extended objects

Center of mass of  $n$  objects considered as point masses:

$$\vec{r}_{CM} = \frac{\sum m_i \vec{r}_i}{\sum m_i}; \quad \sum m_i = M$$



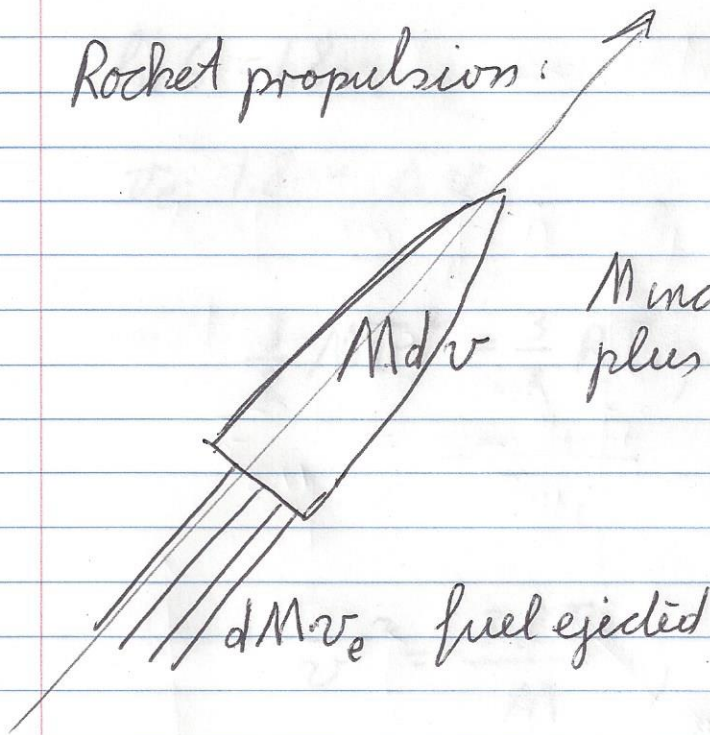
$$M \vec{r}_{CM} = \sum m_i \vec{r}_i$$

$$\frac{d}{dt} \rightarrow M \vec{v}_{CM} = \sum m_i \vec{v}_i \quad ||$$

$$\frac{d}{dt} \rightarrow M \vec{a}_{CM} = \sum m_i \vec{a}_i = \sum \vec{F}_i$$

-p2-

Rocket propulsion:



M includes the mass of the hull  
plus the mass of the fuel  
 $H_2 + O_2$

$$\sum m_i \vec{v}_i = \vec{0} \text{ if in center mass}$$

$v_e$  is constant

$$-v_e dM + M dv = 0$$

$$v_e \frac{dM}{M} = dv$$

$\int_i^f$

$$-v_e \frac{|dM|}{M} = dv$$

interpret  $dM$  as  $-|dM|$

$$-v_e \ln \frac{M_f}{M_i} = v_f - v_i$$

$$v_e \ln \frac{M_i}{M_f} = v_f - v_i \quad \frac{M_i}{M_f} = 6$$

$-p^3$

$$\ln 6 = 1.8$$

$$v_e: 1.8 = \Delta v$$

$$\frac{3}{2} M \bar{v}^2 = \frac{3}{2} RT; \quad M_{\text{water}} = 18 \text{ g} \\ = 0.018 \text{ kg}$$

$$T \approx 2000 \text{ K}$$

$$\bar{v}^2 = \frac{3RT}{M}, \quad \bar{v} \approx \sqrt{\frac{3RT}{M}}$$

$$= \sqrt{\frac{3 \cdot 8.314 \cdot 2000}{0.018}} \approx 1.7 \frac{\text{km}}{\text{s}}$$

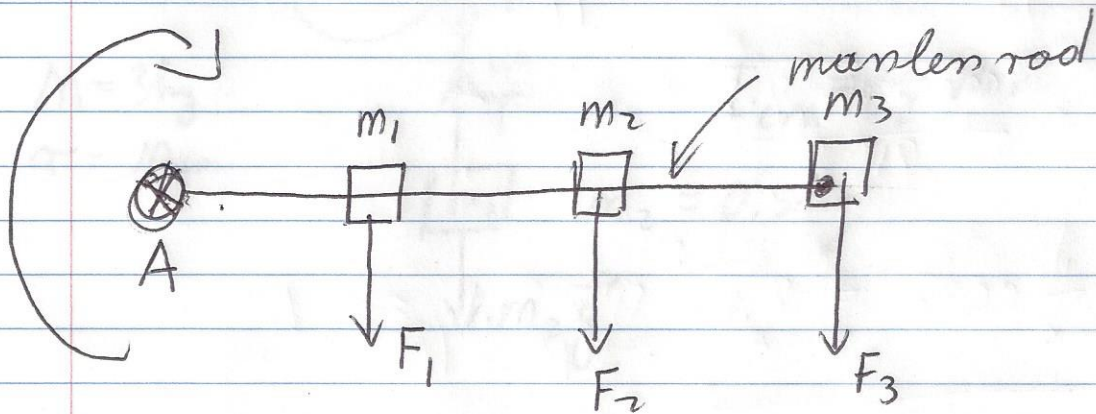
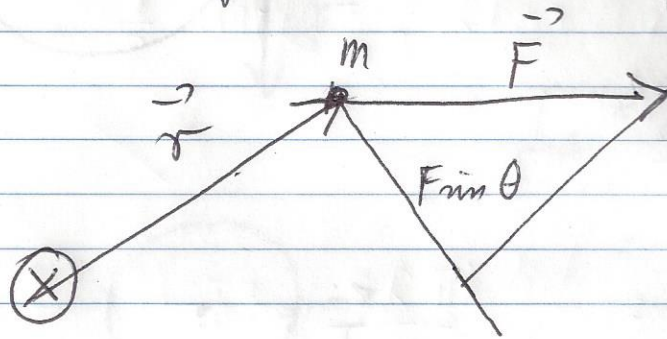
$$\Delta v \approx 1.7 \cdot 10^3 \cdot 1.8 = 3000 \frac{\text{m}}{\text{s}}$$

$$v_{\text{esc}} = 1.4 \sqrt{9.8 \cdot 6.37 \cdot 10^6} = 1.1 \cdot 10^4 \frac{\text{m}}{\text{s}}$$

### Rotational dynamics:

$$\vec{\tau} = \vec{r} \times \vec{F}; \quad \tau = r \cdot F \cdot \sin \theta = r_{\perp} F = r F_{\perp}$$

with respect to a center of rotation.



$$\begin{aligned} \tau_1 + \tau_2 + \tau_3 &= \\ &= r_1 F_1 + r_2 F_2 + r_3 F_3 \quad F = ma \\ &= r_1 m_1 a + r_2 m_2 a + r_3 m_3 a \\ &= r_1 m_1 \alpha r_1 + r_2 m_2 \alpha r_2 + r_3 m_3 \alpha r_3 \\ &= \left( \sum m_i r_i^2 \right) \alpha = I_A \cdot \alpha \end{aligned}$$